

In the Claims:

1. (Currently amended) An apparatus for producing water on
board of an aircraft while using one or more fuel cells,
wherein a partial or complete integration of a water
production unit in the form of one or more comprising at
least one high temperature fuel cells (7) cell that has an
anode side and a cathode side and that is integrated into
a heat-producing arrangement of an aircraft engine, wherein
the heat-producing arrangement optionally additionally
includes at least one combustion chamber, wherein the fuel
cell is adapted to carry out a fuel cell process and the
optional combustion chamber is adapted to carry out a
combustion process, is provided in such a manner that the
combustion chambers (7A) of the aircraft engine are
replaced completely or partially by the high temperature
fuel cells (7) and thus either supplementing or completely
replacing the process that takes place in the conventional
type combustion chambers, characterized in that:

the high temperature fuel cell is an cells (7) are
constructed as type oxide ceramic fuel cell (SOFC - solid
oxide fuel cell), or [[as]] a molten carbonate fuel cell
(MCFC), or a [[type]] fuel cell that is comparable in has a
power and temperature level equivalent to an oxide ceramic
fuel cell or a molten carbonate fuel cell;

[[that]] the apparatus includes a hydrogen supply that
is arranged and adapted to supply pure hydrogen is supplied

26 to the anode side of said high temperature fuel cells (7),
27 that cell;

28 an air intake is arranged and adapted to supply air
29 ~~is supplied~~ to the cathode side of said high temperature
30 fuel cells, that cell;

31 the hydrogen supply and the air intake are further
32 arranged and adapted to supply a mixture of hydrogen and
33 ~~air is supplied~~ to the combustion chambers (7A), that
34 chamber;

35 at least the hydrogen supply is constructed for a
36 closed loop control or can be shut off ~~completely, and that~~
37 completely; and

38 the apparatus further includes a single stage or
39 multistage turbine (16) [[is]] connected downstream of the
40 anode side of the high temperature fuel cell, said turbine
41 adapted to convert ~~converting~~ the thermal energy of [[the]]
42 anode exhaust gas (35) into rotation energy.

- 1 2. (Original) The apparatus of claim 1, characterized in that
2 the conversion of the thermal energy takes place by a
3 Stirling motor and/or by one or more combinations of
4 different thermal engines, for example a turbine and a
5 Stirling motor.

Claims 3 to 24 (Canceled).

1 25. (Previously presented) The apparatus of claim 1, comprising
2 a compressor (13) and means for supplying gained mechanical
3 energy to said compressor.

1 26. (Previously presented) The apparatus of claim 25, wherein
2 said compressor is used for charging said anode side of
3 said high temperature fuel cell (7) with hydrogen (15)
4 under pressure.

1 27. (Previously presented) The apparatus of claim 1, further
2 comprising a condensation process (18) connected downstream
3 of said high temperature fuel cell or high temperature fuel
4 cells (7), said condensation process condensing water out
5 of a portion of anode exhaust gas (35) of said fuel
6 cell (7).

1 28. (Previously presented) The apparatus of claim 1, wherein
2 said high temperature fuel cells (7) are constructed for
3 pressurizing the air or oxygen side, and the fuel or
4 hydrogen side, whereby equal or different pressures are
5 permissible on the anode side and on the cathode side.

1 29. (Previously presented) The apparatus of claim 1, comprising
2 a source (1) for supplying liquid or gaseous hydrogen.

1 30. (Previously presented) The apparatus of claim 29, further
2 comprising a liquid hydrogen evaporator (17) positioned

3 upstream of said high temperature fuel cells (7) or said
4 combustion chambers (7A).

1 31. (Currently amended) The apparatus of claim 30, further
2 comprising an anode ~~exhaust~~ exhaust gas condenser (18) and
3 wherein said evaporator (17) is constructed to be operable
4 by process heat of said anode exhaust gas condenser (18).

1 32. (Previously presented) The apparatus of claim 31, wherein
2 said evaporator (17) is constructed as a pipe bundle heat
3 exchanger which is arranged as a ring shape around said
4 condenser (18) or circularly within said condenser (18).

1 33. (Previously presented) The apparatus of claim 31, wherein
2 at least a portion of said condenser (18) is operable with
3 cooling air (19).

1 34. (Previously presented) The apparatus of claim 1, further
2 comprising a container (32) for collecting used water and
3 not needed condensate as gray water.

1 35. (Currently amended) The apparatus of claim 1, further
2 comprising a gray water evaporator (33) wherein air (20)
3 heated in a condensation process is used for evaporating
4 gray water, a pump [(37)] (45) for feeding gray water
5 into said evaporator (33), and a filter provided for

6 retaining solid and suspended matter out of said gray
7 water.

1 36. (Previously presented) The apparatus of claim 1, wherein
2 produced steam is blown in upstream of a second turbine
3 stage (9) of said multistage turbine where said steam is
• 4 mixed with cathode exhaust air.

1 37. (Previously presented) The apparatus of claim 1, wherein
2 any germs and microorganisms present in gray water (32) are
3 thermally killed.

1 38. (Previously presented) The apparatus of claim 1, further
2 comprising means for withdrawing water of distilled quality
3 from a condensation process (18) and for distributing said
4 distilled quality water, a salination station (23) for
5 adding a dose of salt to produce drinking water for
6 galleys, hand wash basins and showers and for supplying
7 distilled water to toilets and humidifiers.

1 39. (Previously presented) The apparatus of claim 1, wherein
2 said multistage turbine comprises turbine stages (8, 9) for
3 driving compressor stages (5, 6) and a fan (11), and
4 wherein compressor stages (5, 6) pressurize an air side of
5 said high temperature fuel cells (7) and of said combustion
6 chambers (7A).

1 **40.** (Previously presented) The apparatus of claim 39, wherein
2 an air throughput (3) of said fan (11) is used either in an
3 engine for propulsion or in an APU for pressurization of
4 pressurized air systems and/or of an air conditioning
5 system.

1 **41.** (Previously presented) The apparatus of claim 39, wherein
2 said fan (11) is coupled with a first compressor stage (6)
3 and with the second turbine stage (9), and wherein a second
4 compressor stage (6) and the first turbine stage (8) are
5 coupled with each other and run on coaxial shafts with
6 different revolutions per minutes.

1 **42.** (Previously presented) The apparatus of claim 41, wherein
2 the number of coupled compressor stages and turbine stages,
3 the direction of rotation of these stages, and the number
4 of coaxial shafts rotating one within the other are
5 constructed at discretion.

1 **43.** (Previously presented) The apparatus of claim 1, further
2 comprising a waste water collection tank (28) for
3 collecting waste water and wherein said waste water is
4 completely or partially dehydrated at (30) and the thus
5 gained water portion is fed into a gray water collection
6 tank (32).

1 45. (Previously presented) The apparatus of claim 1, wherein
2 said combustion chambers and said high temperature fuel
3 cells are operable separately and in any desired
4 combination.

1 46. (Previously presented) The apparatus of claim 1, wherein
2 individual combustion chambers or high temperature fuel
3 cells are adapted to be switched off for a separate
4 operation of combustion chambers or high temperature fuel
5 cells.